Hybrid Oxidizer Tank Filling and Dynamics

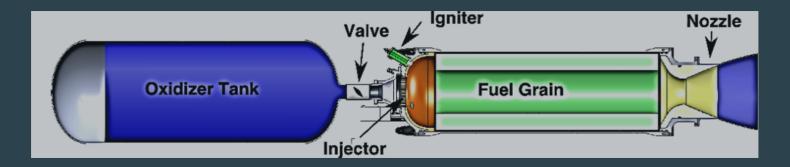
Team #3 April 25, 2020 Advisor: Dr. Jeffrey Santner

Agenda

- Introduction
- Python Simulations
- ANSYS Simulations
- Structural Design & Analysis
- Control Systems
- Conclusion

Introduction

- Sub-unit of the Eagle Rocketry SEDS Club at Cal State LA
 - Aerospace centric club researching rocket engines: Hybrid
 - SEDS: Students for Exploration and Development of Space



Objectives

- Remotely transfer 15kg of nitrous oxide from a commercially purchased vessel
- Create a testing structure for the nitrous oxide pressure vessel which can double as a launch pad
- Predict the time taken for the filling process



Meet the Team



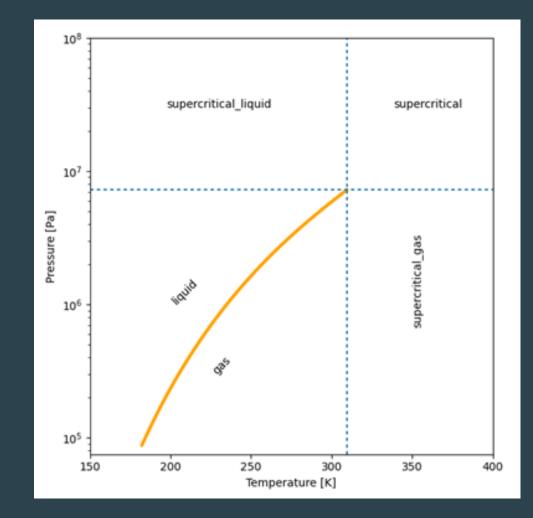
Orifiel Ortiz



Python Simulations

Model Overview

- Based on a modified Equilibrium Model. [1]
 Limitations:
- CoolProp cannot compute thermodynamic properties when the fluid is supercritical
- CoolProp cannot calculate N₂O thermal conductivity or viscosity at any temperature or pressure



Thermal Conductivity and Viscosity

 Supercritical Thermal Conductivity is calculated from tables published by Richter and Sage. [2]

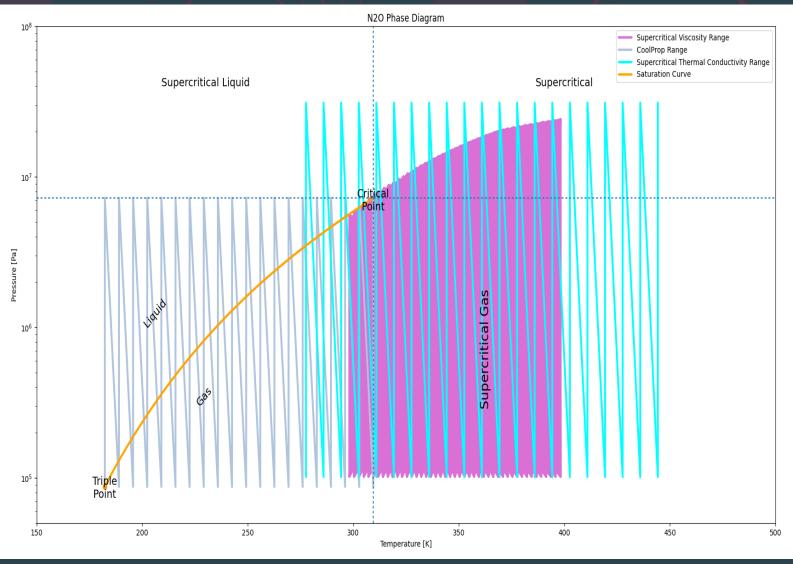
Validity range: 277.59*K* - 444.26*K* 101.33*kPa* - 34.47 *MPa* Supercritical Viscosity is calculated from work by Takashi et. al. [3]

Validity range: 298.15*K* – 398.15*K* 102.50*kPa* – 5.50*MPa*

Thermal Conductivity and Viscosity

 Saturation values calculated from equations published by ESDU. [4]

Validity range: 182.33*K* - 309.57*K* 87.3*kPa* - 7.24*MPa*

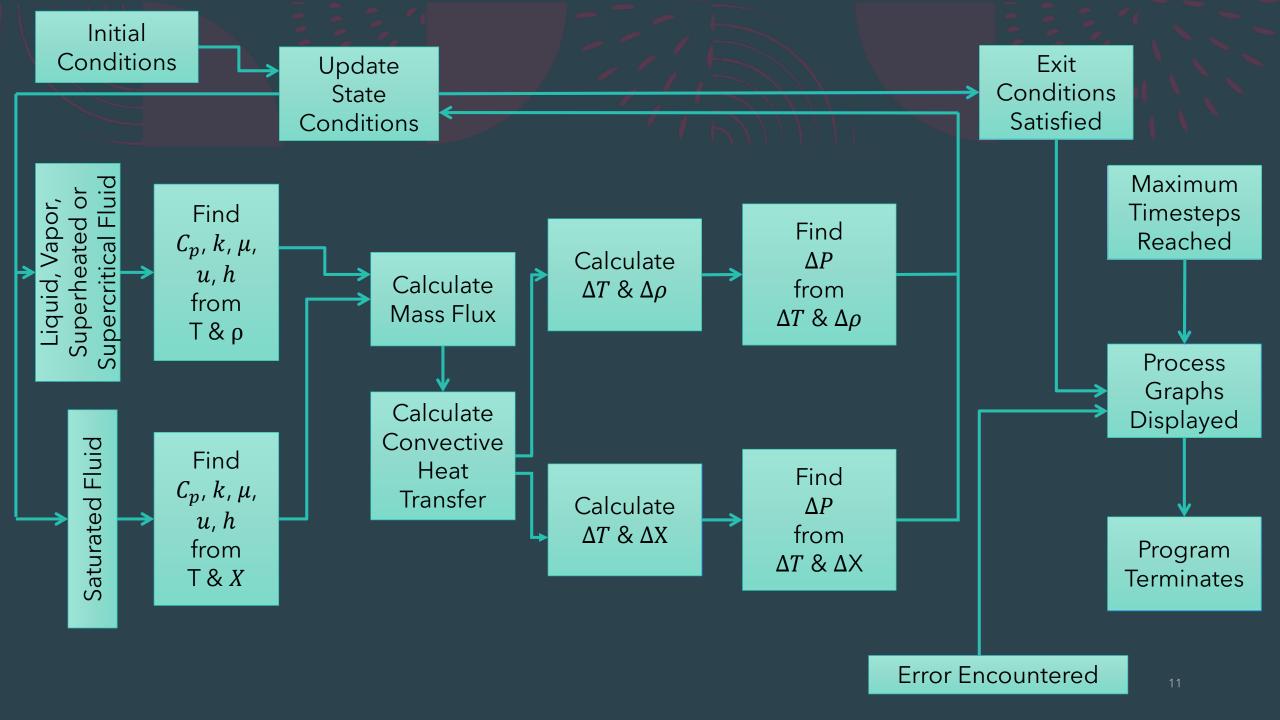


Solution

 Helmholtz Functional Equations using temperature and density as the two known supercritical fluid states. [5]

$$\alpha(\delta,\tau) = \alpha^0(\delta,\tau) + \alpha^r(\delta,\tau)$$

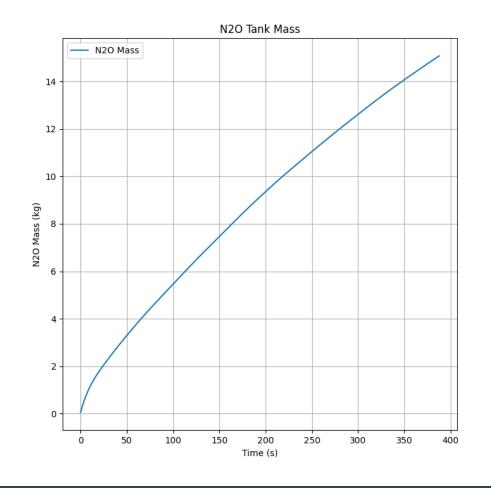
- $\alpha(\delta, \tau)$ = Helmholtz Energy
- δ = normalized density
- τ = normalized temperature
- Other fluid properties can be calculated through the First Law of Thermodynamics, with a known Helmholtz Energy

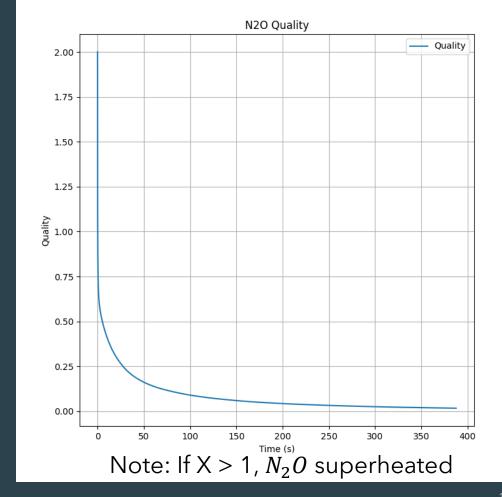


Boundary Conditions

- Initial/Atmospheric Temperature: 298K
- Initial Pressure: air at 101.33 kPa
- Tank Length: 1.27 m (50 in)
- Tank Inner Diameter: 152.4 mm (6 in)
- Source Tank Pressure: 689.48 kPa (1000 Psi)
- Total Tank Mass: 15kg

m & *X*

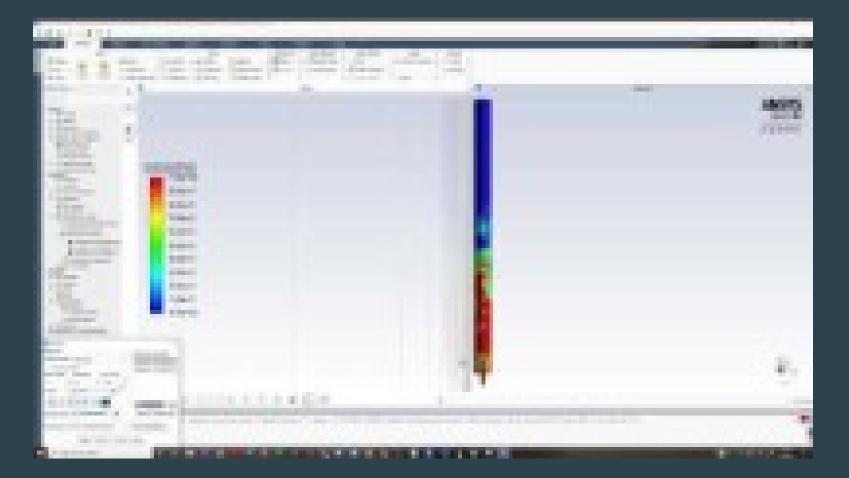




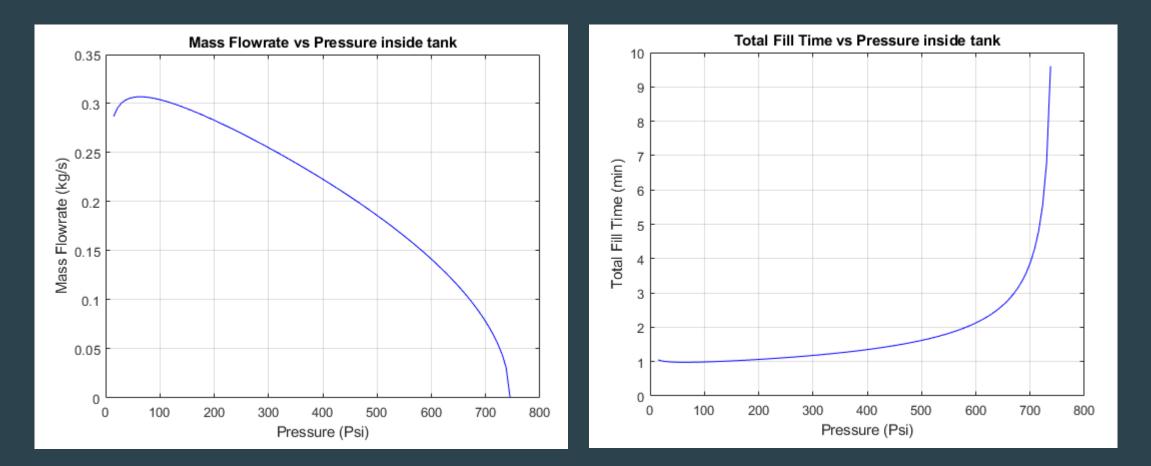


Ansys Pressure Vessel Simulations

Oxidizer Tank Filling



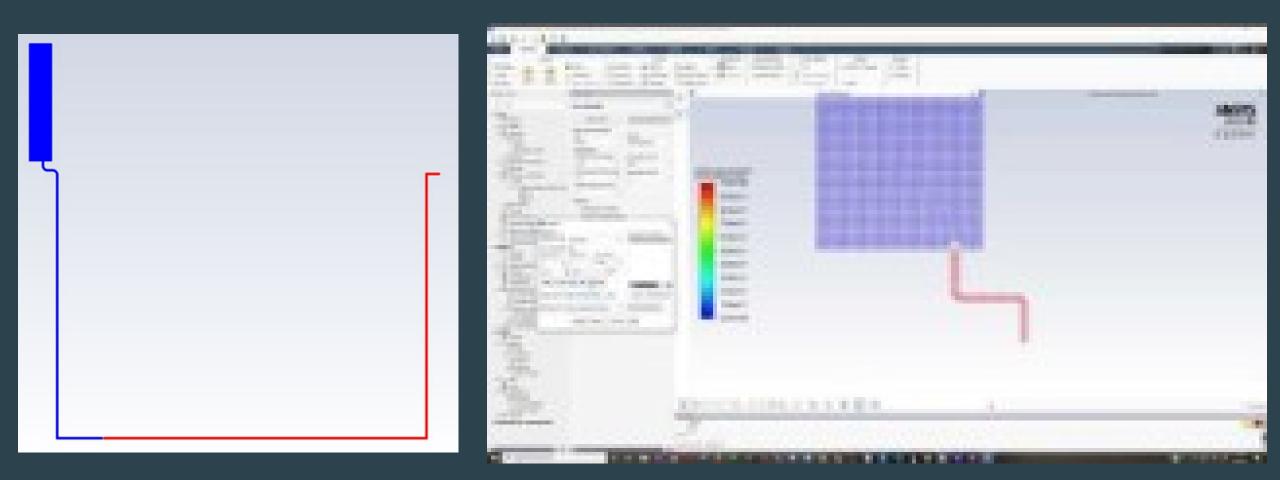
MATLAB



Ground Fill System

- Inlet diameter to Oxidizer Tank: 0.25in
- Inner diameter of piping system is 0.25in
- Going to open the source tank at varying pressures to slow down the filling time

Ground Fill System



Mass Requirements

- Do we have enough N20 to fill the tank and for the ground system piping?
 - N2O mass system = 28kg
 - N2O source tank = 27kg
 - 2 source tanks from vendor

Grade & Gas Specification	Cyl. Size	Product Code�	Contents Pounds	Cylinder Rental	Cylinder Pressure (psi)	CGA Outlet	Nominal Cylinder Dim. (in.)�	Shipping Wt. Lbs�
Nitrous Oxide	200	NO AA200	60	YES	745	326	9 x 55	202
AA	G	NO AA56	56	YES	745	326	8.5 x 55	170
99.6%	G	NO AA50	50	YES	745	326	8.5 x 55	164

F

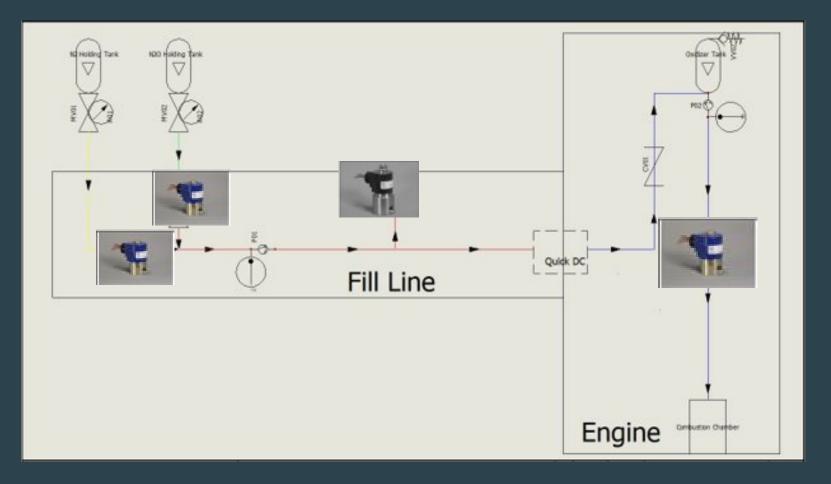
Ansys & Valve Simulations



Amy Moore

•	Piping	&	Va	lve	F	lui	C
	Analys	t					

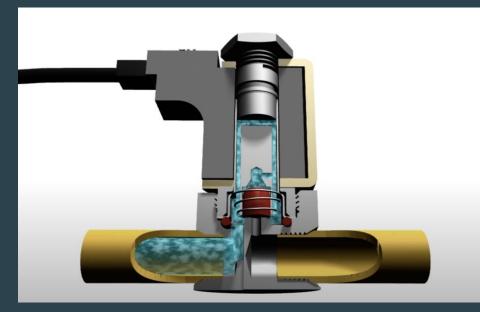
Schematic of Valve Layout

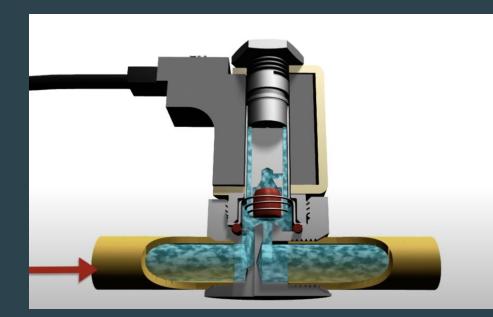


V01-Normally Closed (Blue) V02-Normally Closed V03-Normally Open (Gray) V04-Normally Closed

Valve System Analysis

- CFD simulations performed on valves
- ANSYS and SimScale software was implemented
- Valve normally closed tested
- Inner and outer diameter 0.25 in.

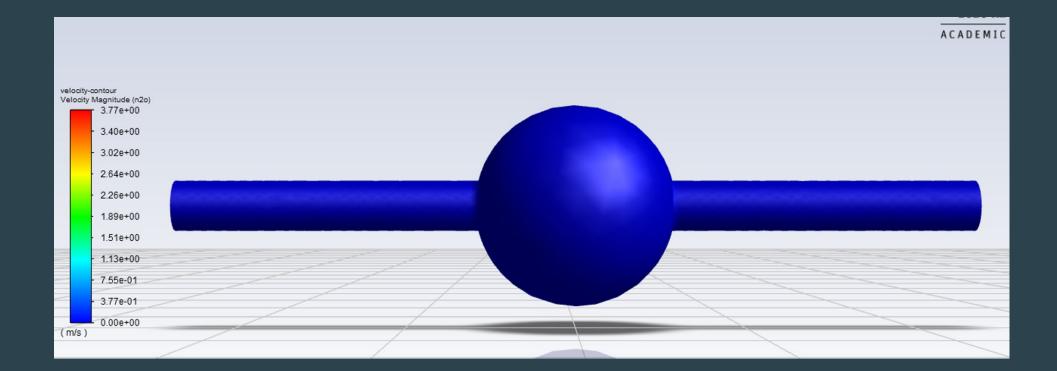




Valve Analysis ANSYS

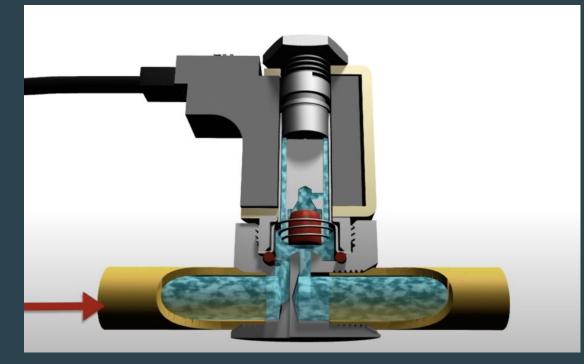
• Velocity entering valve 3.77 m/s

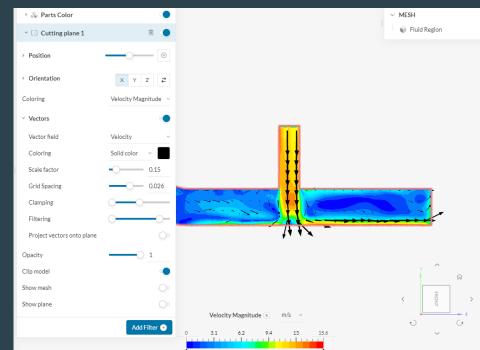
Mass flow rate entering valve 3 kg/s



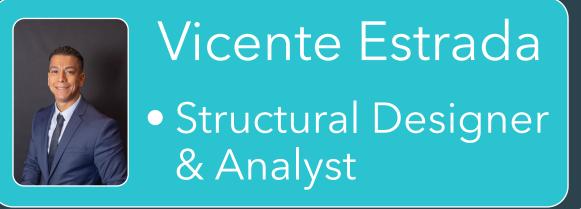
Valve Analysis SimScale

SimScale is a program able to understand CAD models
Velocity at the stopper region is 9 m/s





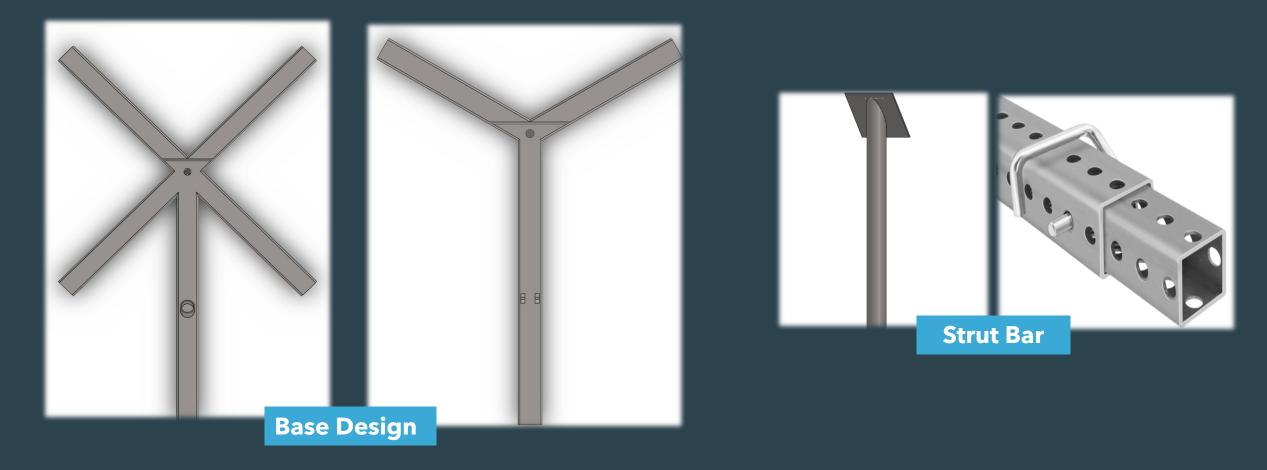
Base Structure Analysis



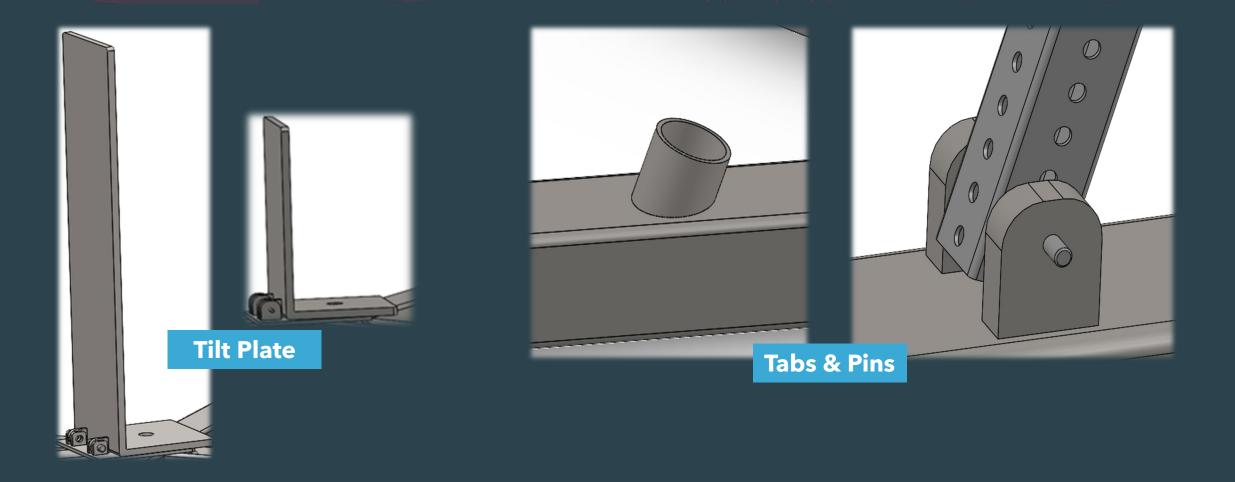
Base Structure



Major Changes



Major Changes



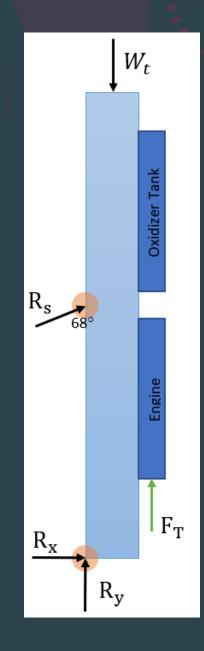
Hand Calculations

 $W_T = 284.78N$ $F_T = 2500N$

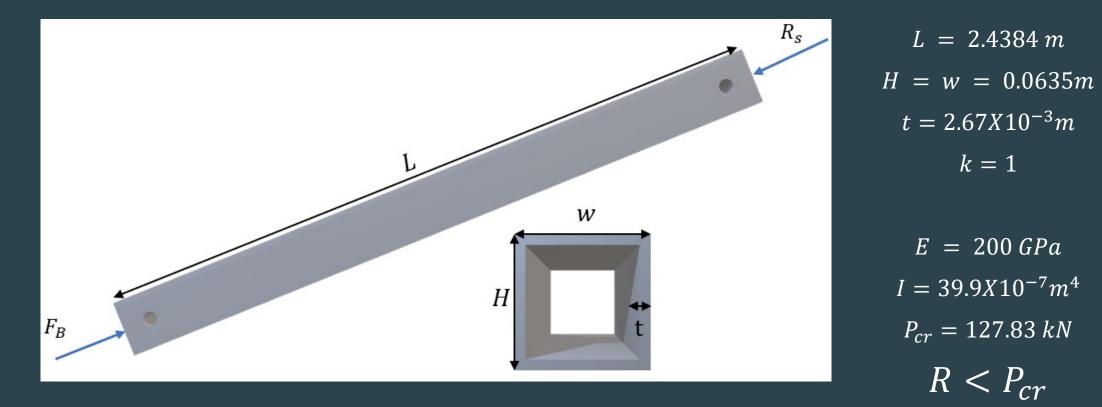
$$\Sigma F_{y}: 0 = R_{s} Cos(68) + R_{y} + F_{T} - W_{T}$$
$$\Sigma F_{x}: 0 = R_{s} Sin(68) + R_{x}$$
$$\Sigma M: 0 = (0.3048m)F_{T} - (2.231m)R_{s} Sin(68) - 21.7Nm$$

$$R_y = -2349.28N$$

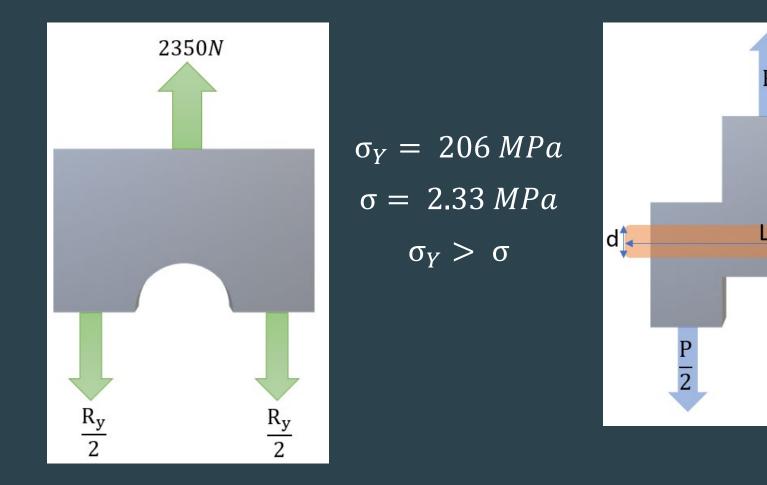
 $R_x = -331.82N$
 $R_s = 357.88N$



Hand Calculations



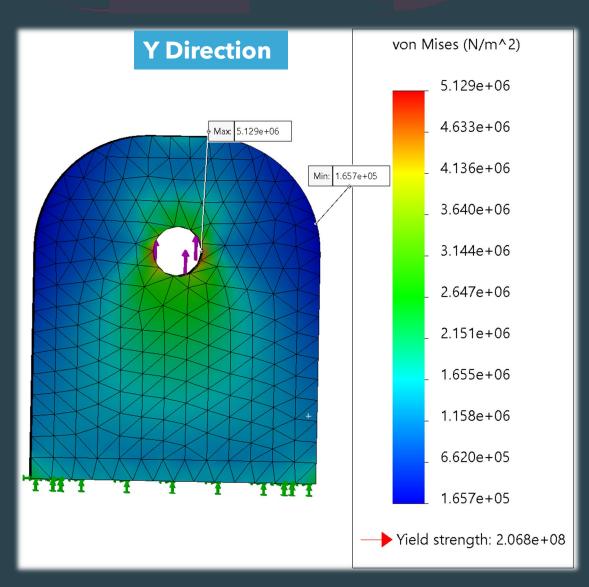
Hand Calculations

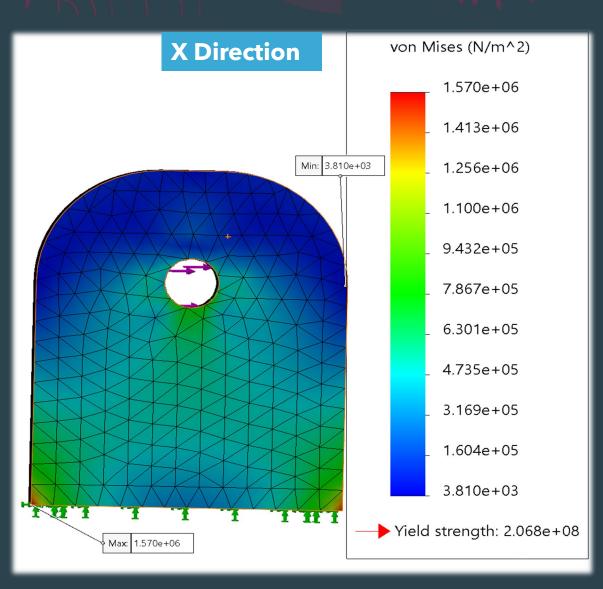


 $P = R_y$ $\tau_U = 66278.5N$ L = 0.127md = 0.4375m

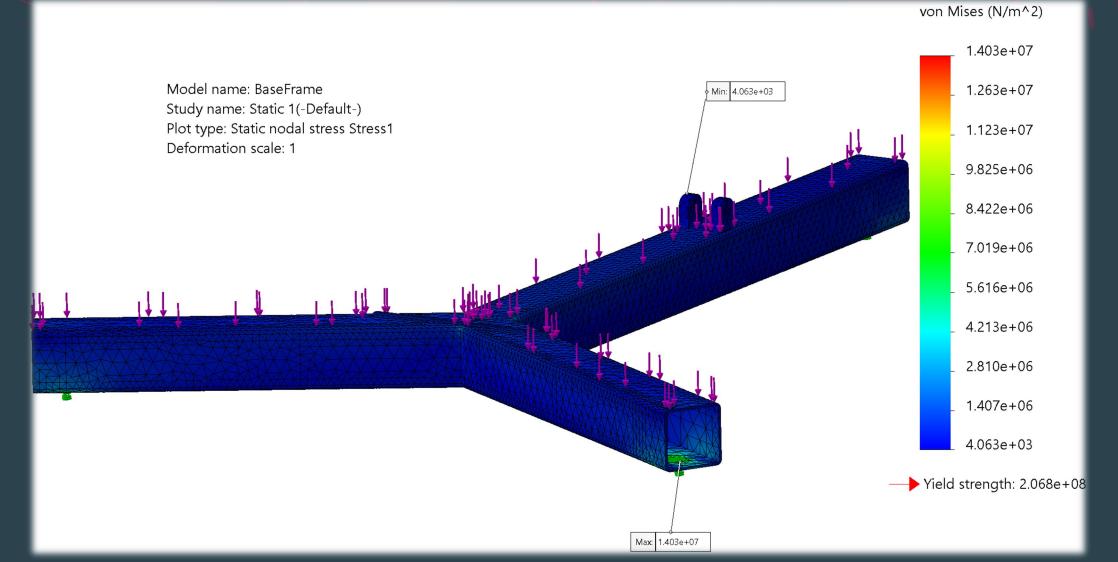
Stainless Steel Pin

FEA



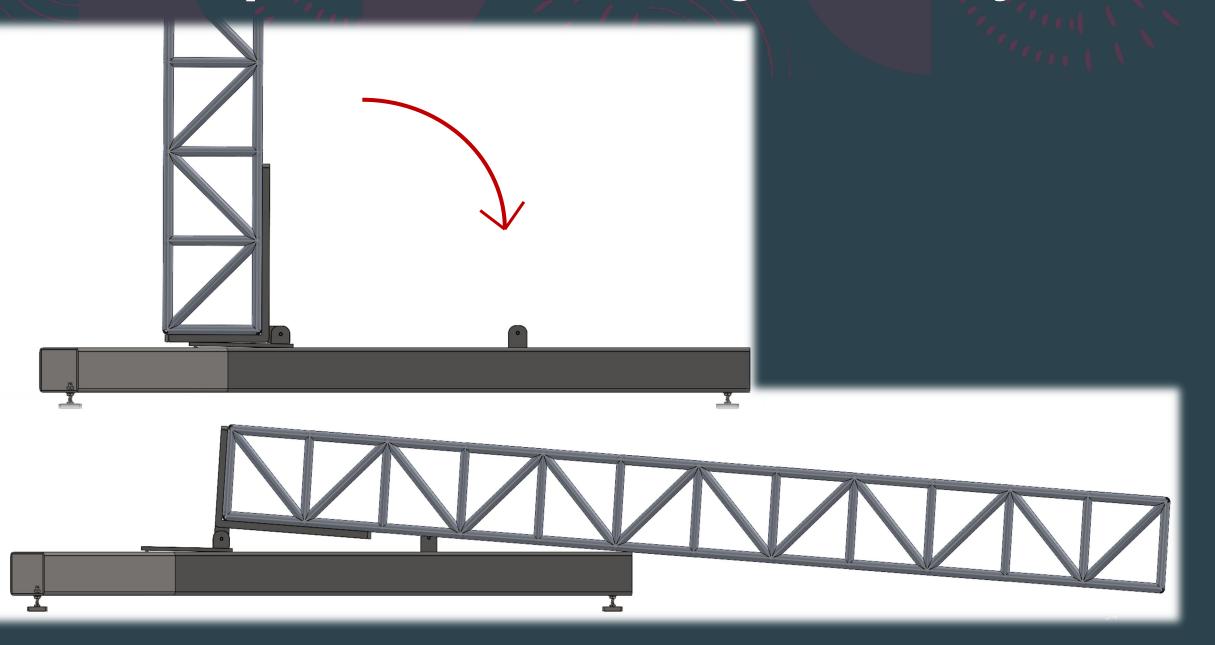


FEA



3

Transportation/Mounting Assembly





Test Stand Analysis

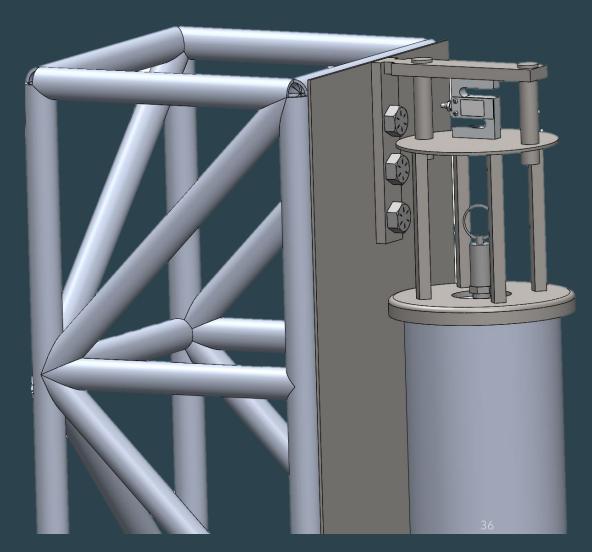


Alex Guilbaud

 Structural Designer & Analyst

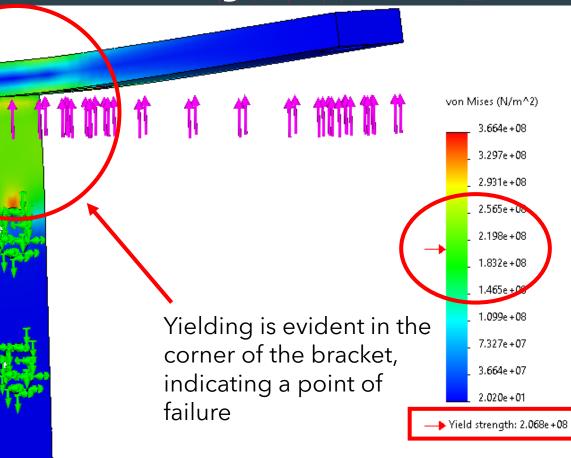
Mechanical Force Analysis

- Concern:
 - Original bracket has potential to fail
- Resolve:
 - Calculate bending stresses and conduct FEA; determine if true



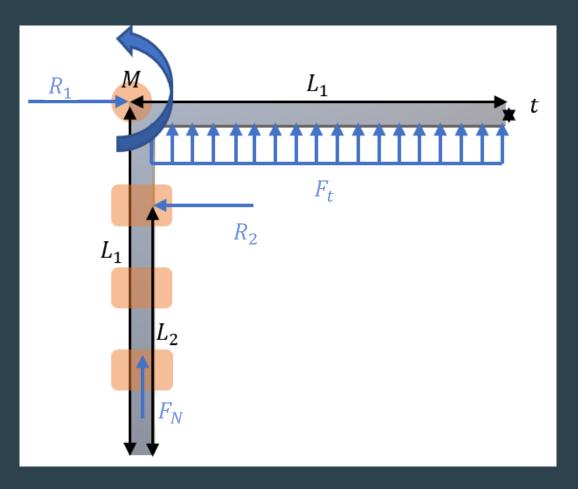
Mechanical Force Analysis

Static fire top bracket v2.SLDPRT	Optio	ns	
Override Mass Properties	Recalculate		
Include hidden bodies/components			
Create Center of Mass feature			
Show weld bead mass			
Report coordinate values relative to:	default	\checkmark	
Mass properties of Static fir	e top bracket v2		
Configuration: Default			
Coordinate system: de	efault		
Density = 0.29 pounds per	cubic inch		
Mass 276 pounds			
Mass = 2.76 pounds			
Volume = 9.55 cubic inches			
- startie = 5155 cubic merica			
Surface area = 60.07 square	e inches		
Center of mass: (inches)			
X = 2.07			
Y = 0.74			
Z = -5.53			
Principal avos of inortia and	principal moments of	inertia: (pounds * square in	cho
Taken at the center of mass		inertia. (pounds square in	che
Ix = (-0.71, 0.00, 0.71)			
$I_{x} = (-0.71, -0.00, -0.71)$ $I_{y} = (-0.71, -0.00, -0.71)$	Py = 24.00		
z = (0.00, 1.00, 0.00)	Py = 24.00 Dz = 29.55		
12 - (0.00, 1.00, 0.00)	FZ = 23.35		
Moments of inertia: (pound	ds * square inches)		
Taken at the center of mass		output coordinate system.	
Lxx = 15.37	Lxy = 0.00	Lxz = -8.68	
Lyx = 0.00	Lyy = 29.55	Lyz = 0.00	
Lzx = -8.68	Lzy = 0.00	Lzz = 15.27	
	,		
Moments of inertia: (pound	ds * square inches)		
Taken at the output coordin	nate system.		
Ixx = 101.14	lxy = 4.21	lxz = -40.31	
lyx = 4.21	lyy = 125.70	lyz = -11.21	
	lzy = -11.21	lzz = 28.63	



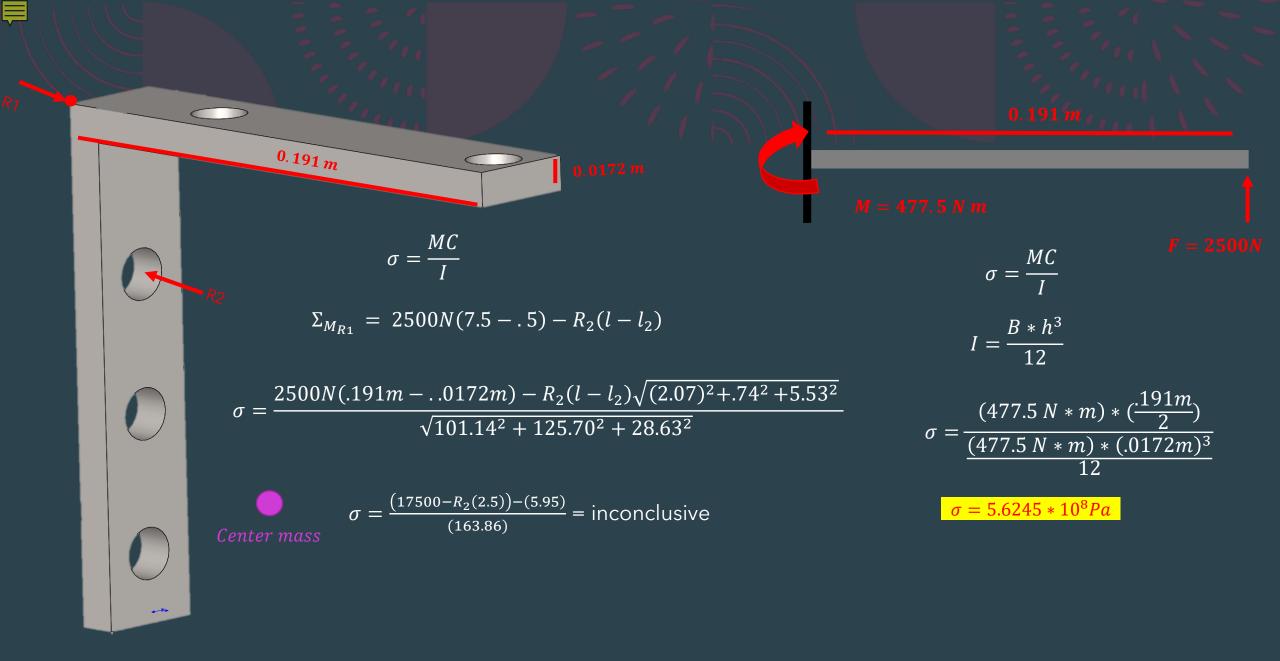
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Hand Calculations



 $L_1 = 0.191m$ $L_2 = 0.127m$ t = 0.0127m

 $F_T = 2500N$ $\Sigma F_x: R_1 = R_2$ $\Sigma F_y: F_n = -mg + F_T$ $\Sigma M_{R_1}: M = 17500Nm - R_2(L_1 - L_2)$

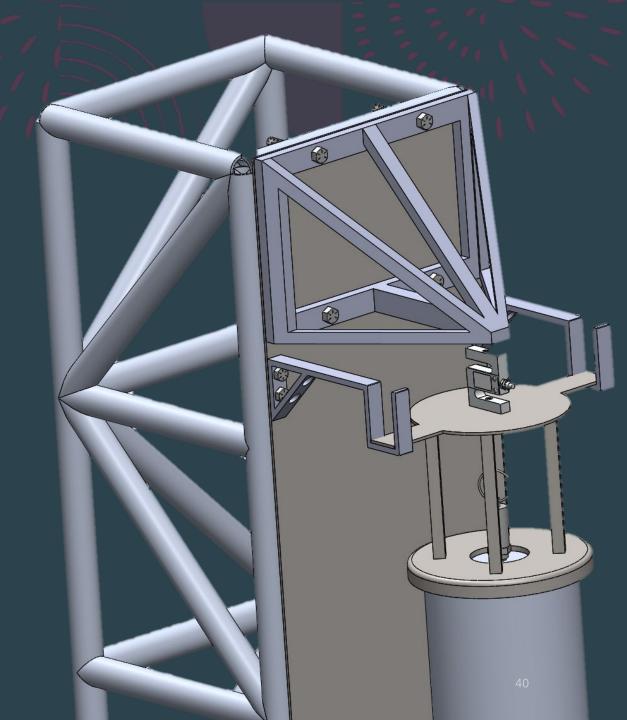


Design Changes

 Evidence of failure present; redesign necessary

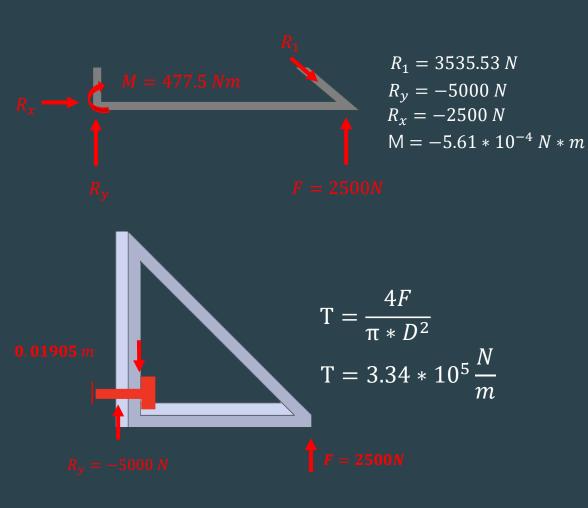
 Reinforced truss shape adopted to prevent bending

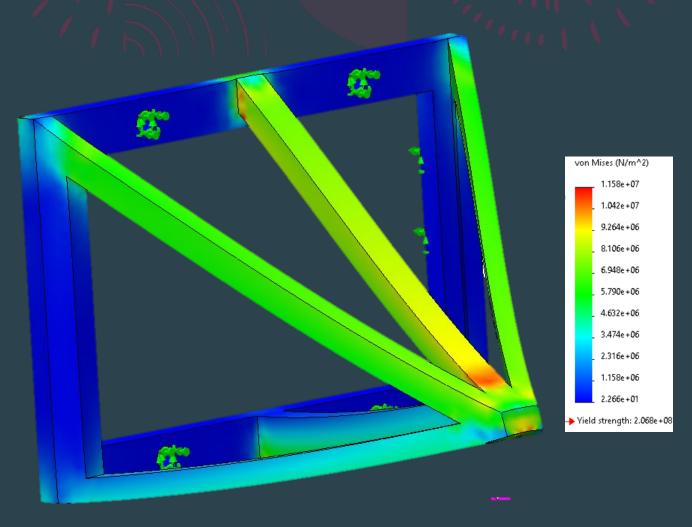
 Water cut in 4 separate pieces and welded together



Bracket FEA

- no yielding is evident with new design
- Same mounting hardware can be used

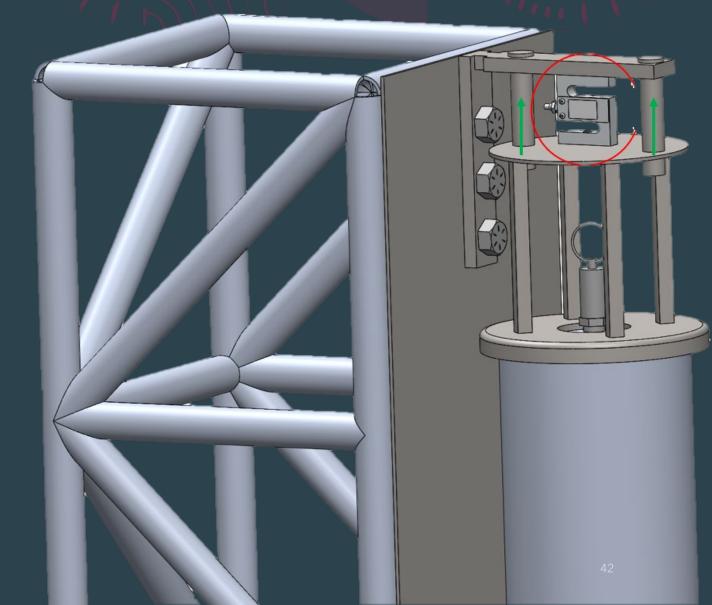


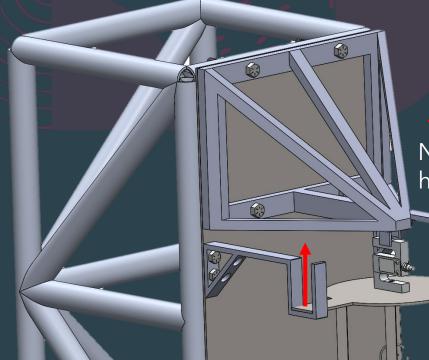


Design changes

• Original slide system for load cell could bind/stuck during test fire

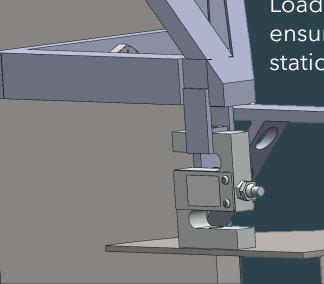
 Bracket redesign meant redesigning this slide system as well





New slide system doubles as a hanger for motor assembly

Load cell is guided with tabs to ensure constant contact with static fire bracket



Controls System



Anthony UrzuaControls System Designer

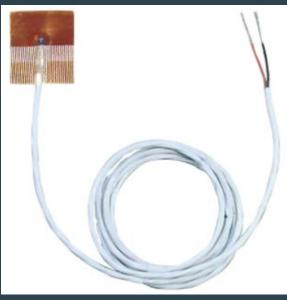
Control System Design

 Design a GUI to allow operators to control the fill, drain, and purging sequencing of the oxidizer tank using LabVIEW and Arduino

Selected Sensors

- Explosion-Proof Industrial Pressure Transducer
- Self-Adhesive Polyimide Fast Response Thermistor Surface Sensor





Issues

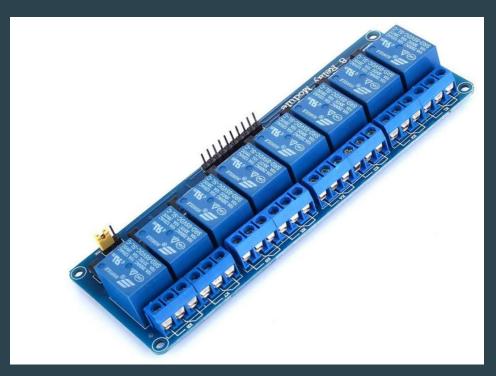
 Pressure transducer was selected but was not able to be ordered. Could not get a hold of any technical documentation.

Additional Hardware

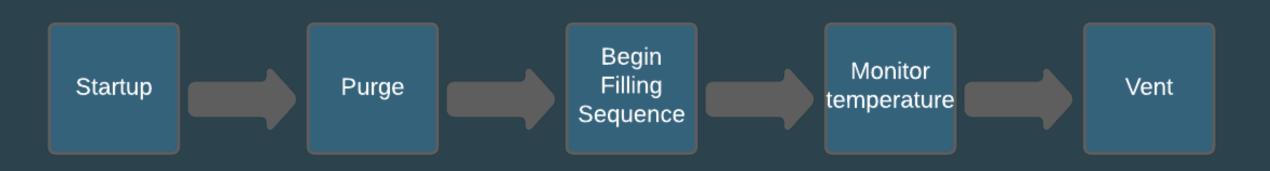
• 6 V Battery

• 8-Channel 5V DC Relay Module Control Board

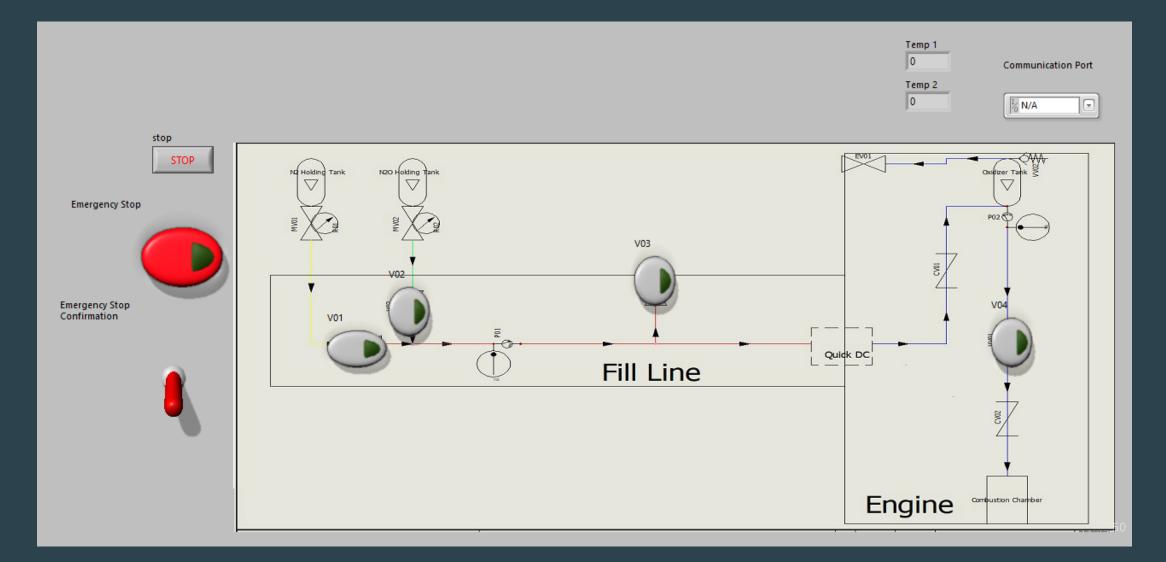




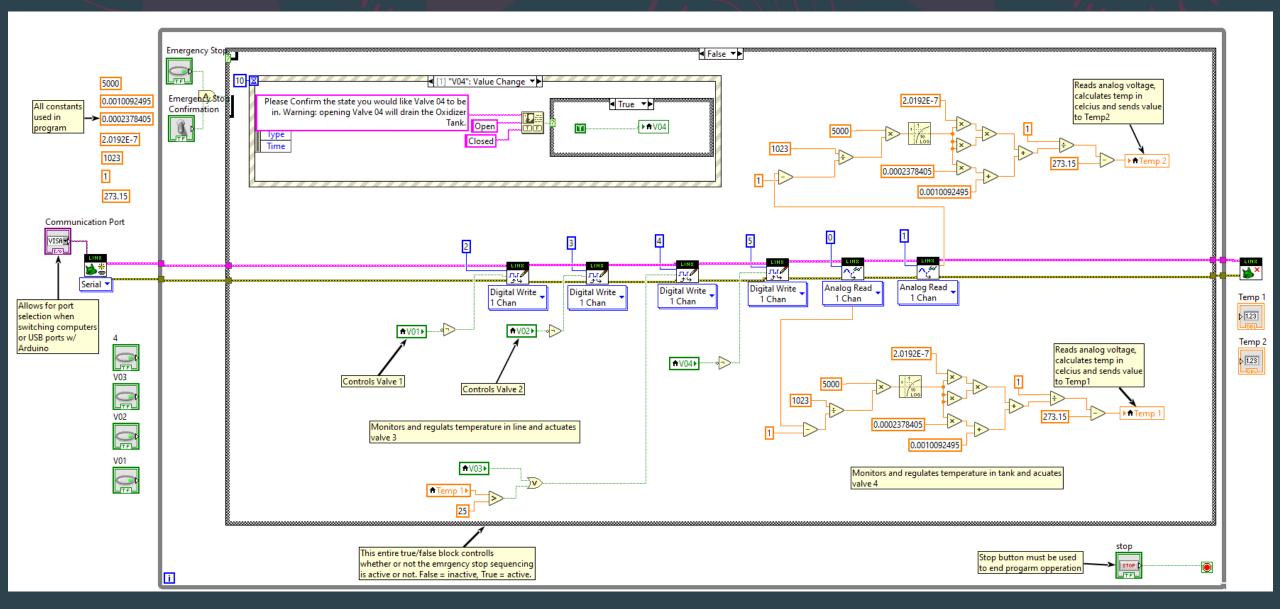
Current Block Diagram



Front Pannel

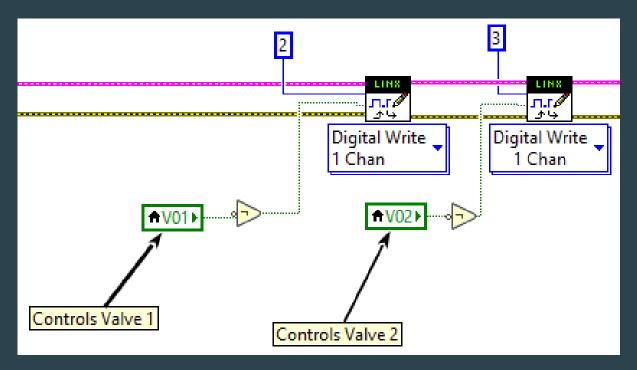


Block Diagram



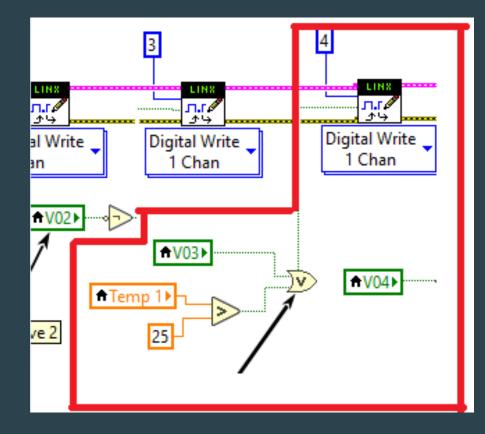
Valve actuation logic

Normally Closed:

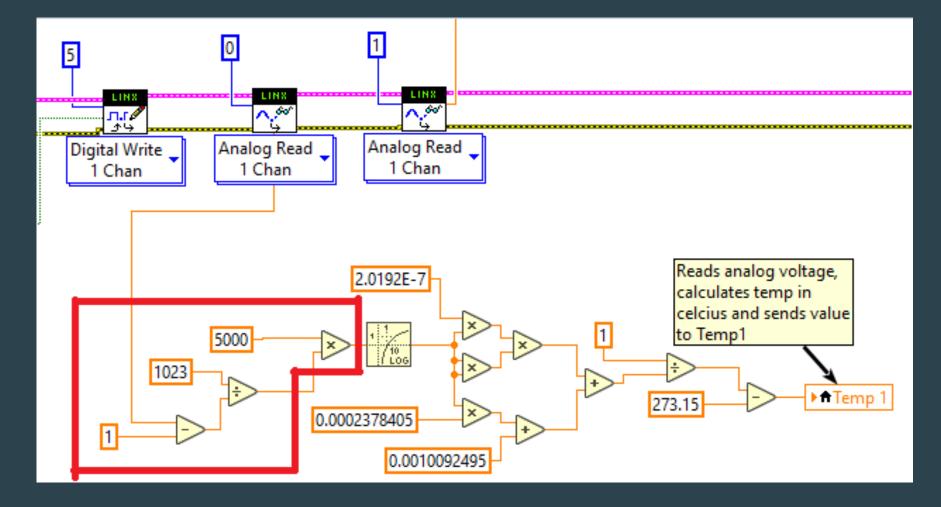


Valve Actuation logic

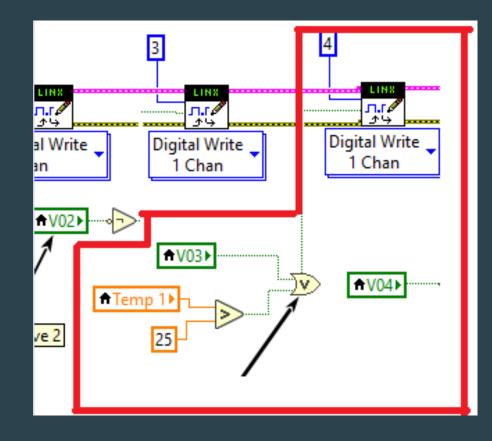
Normally Open:



Temperature Calculation



Temperature Regulation



Truth Table:

Temp	Valve	Temp	Valve
	controller	Valve	Position
0	0	0	Closed
0	1	1	Open
1	0	1	Open
1	1	1	Open

In Conclusion

- Python and ANSYS total times are within 2 minutes of each other.
- Valve losses increase total fill time.
- Structural designs reduce manufacturing complexity.
- Structure can withstand static fire forces.
- Controls system allows process to be done autonomously.

Thank you!

- Dr. Santner
- ERSEDS
- GC Valves
- EMJ Metals
- College of ECST

References

- [1] J. E. Zimmerman, B. S. Waxman, B. J. Cantwell and G. G. Zilliac, "Review and Evaluation of Models for Self-Pressurizing Propellant Tank Dynamics," in *49th AIAA/ASME/SAE/ASEE Joint Propulsion Conference*, San Jose, 2013.
- [2] N. G. Richter and B. H. Sage, "Thermal Conductivity of Fluids. Nitrous Oxide," *Journal of Chemical & Engineering Data*, vol. 8, no. 2, pp. 221-225, 1963.
- [3] M. Takahashi, N. Shibasaki-Kitakawa, C. Yokoyama and S. Takahashi, "Viscosity of Gaseous Nitrous
 Oxide from 298.15 K to 398.15 K at Pressures up to 25 MPa," *Journal of Chemical & Engineering Data,* vol. 41, no. 6, pp. 1495-1498, 1996.
- [4] ESDU, "Thermophysical properties of nitrous oxide," *ESDU Series on Physical Data, Chemical Engineering,* September 1991.
- [5] E. W. Lemmon and R. Span, "Short Fundamental Equations of State for 20 Industrial Fluids," *Journal of Chemical Engineering Data*, vol. 51, no. 3, pp. 785-850, 2006.